Fundamentals of Control

State space and Laplace transform

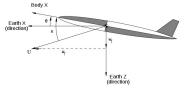
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Notation

- l, d: lift and drag;
- C_l, C_d: non-dimensional lift and drag coefficients;
- L, M, N: roll, pitch, yaw moments;
- C_L, C_M, C_N: non-dimensional coefficients of roll, pitch, yaw moments;
- $C_{l_{\alpha}} = \partial C_l / \partial \alpha$.

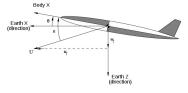
Review: longitudinal and lateral angles





$$\theta = \alpha + \gamma$$

Review: longitudinal and lateral angles





$$\theta = \alpha + \gamma$$

$$\xi = \phi + \beta$$

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(cross-range; heading; roll; spiral; dutch roll mode).

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(cross-range; heading; roll; spiral; dutch roll mode).

Today's objective is to understand the analyse method of the longitudinal dynamics.

$$\begin{array}{rcl} m(\dot{w}-qU_{0}) &=& F_{z}, & (1) \\ && I_{yy}\dot{q} &=& M. & (2) \end{array} \end{array}$$

$$\begin{array}{rcl} m(\dot{w'} - q'U_0) &=& Z^c, & (3) \\ I_{yy}\dot{q'} &=& M_ww' + M_qq' + M^c. & (4) \end{array}$$

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Assume we only have one aerodynamic device: rudder, i.e.

$$\begin{split} \delta Z^{c} &= m Z_{d} \delta r, \\ \delta M^{c} &= I_{vv} M_{r} \delta r. \end{split}$$

$$\begin{array}{lll} m(w'-q'U_0) &=& Z^c, \\ I_{yy}q' &=& M_ww' + M_qq' + M^c. \end{array} (3)$$

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 $\delta Z^{c} = m Z_{d} \delta r,$

$$\delta M^{c} = I_{yy}M_{r}\delta r.$$

How to analysis the dynamics?

To answer this question, we need to use methods in linear system.

Linear dynamical system

Continuous-time linear dynamical system (LDS) has the form

$$\frac{dx}{dt}=A(t)x(t)+B(t)u(t),\quad y(t)=C(t)x(t)+D(t)u(t).$$

Linear dynamical system

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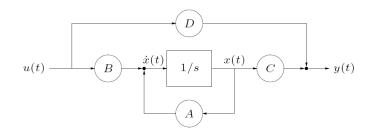
Continuous-time linear dynamical system (LDS) has the form

$$\frac{\mathrm{d}x}{\mathrm{d}t} = A(t)x(t) + B(t)u(t), \quad y(t) = C(t)x(t) + D(t)u(t).$$

If A, B, C, D are time-invariant, we can have the corresponding transfer function:

$$\mathbf{Y}(\mathbf{S}) = \mathbf{G}(\mathbf{S})\mathbf{X}(\mathbf{S}).$$

Block diagram



- 1/s represent integral;
- A_{ij} is gain factor from state x_j into integrator i;
- B_{ij} is gain factor from input u_j into integrator i;
- C_{ij} is gain factor from state x_j into output y_i;
- $\bullet \ D_{ij}$ is gain factor from input u_j into output $y_i.$

Block diagram

Q: Draw block diagram for

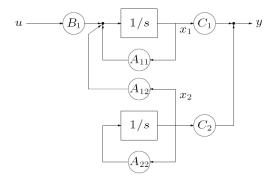
$$\frac{\mathrm{d}}{\mathrm{dt}} \left[\begin{array}{c} x_1 \\ x_2 \end{array} \right] = \left[\begin{array}{cc} A_{11} & A_{12} \\ 0 & A_{22} \end{array} \right] \left[\begin{array}{c} x_1 \\ x_2 \end{array} \right] + \left[\begin{array}{cc} B_1 \\ 0 \end{array} \right] u, \quad y = \left[\begin{array}{cc} C_1 & C_2 \end{array} \right] \left[\begin{array}{c} x_1 \\ x_2 \end{array} \right]$$

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- $\bullet \ x_2$ is not affected by input u, i.e., x_2 propagates autonomously;
- x₂ affects y directly and through x₁.

(Prof. Huang)

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Course introduction

• Prerequisite: Linear algebra, Fourier transform. (Read through the related textbooks yourself again!)

Introduction

System dynamics

- Control systems are dynamic \longrightarrow transient + steady state.
- Transient motion is important. In the case of a lift, slow transient motion makes people impatient, rapid motion is uncomfortable. If the lift oscillates this is disconcerting.
- Steady state is also important, it can have an error, i.e. the step up or down on the exit form a lift.

Introduction

Transfer function

- A mathematical representation of the relation between the input and output of a linear time-invariant system (will be defined later).
- Q: If above system is linear, what is its transfer function?

Introduction

Dynamics

We must have a dynamic system: Total = Natural + Forced ↓ describes how the system dissipates energy This means that the equivalent differential equation equals a homogeneous solution plus a particular solution.

Some LDS terminology

- Most linear systems encountered are time-invariant: A, B, C, D are constant, i.e., don't depend on t.
- When there is no input u (hence, no B or D) system is called autonomous.
- Very often there is no feedthrough, i.e., D = 0.
- When u(t) and y(t) are scalar, system is called single-input, single-output (SISO); when input & output signal dimensions are more than one, MIMO.

State space model and more...

- **①** The above cases are represented by state space models;
- ² They are in the time domain and the bare bone of modern control.

State space model and more...

- **①** The above cases are represented by state space models;
- ⁽²⁾ They are in the time domain and the bare bone of modern control.
- In the very beginning of control (so-called classical control), frequency domain is more preferred.
- **(4)** Time domain is still a part of classical control.
- Next we will present primitive mathematical knowledge (Laplace transform).
- Another reason to study Laplace transform is to achieve transfer function for state space model.

Laplace transform

$$\mathcal{L}[f(t)] = F(s) = \int_{0-}^{\infty} f(t) e^{-st} dt$$

Laplace Integral, Q: why?

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- Conservation;
- Causality:

The relationship between an event (the cause) and a second event (the effect), where the second event is understood as a consequence of the first;

Current state (x(t)) and output (y(t)) depending on past input $(u(\tau) \text{ for } \tau \leq t)$ is causal;

Current state (and output) depending on future input is anti-causal.

Laplace transform

$$\begin{array}{l} \mathbf{\mathcal{L}}[tf(t)] = -F'(s) \\ \mathbf{\mathcal{L}}[e^{-at}f(t)] = F(s+a) \\ \mathbf{\mathcal{L}}[f(t-T)] = e^{-sT}F(s) \\ \mathbf{\mathcal{L}}[f(at)] = \frac{1}{a}F(\frac{s}{a}) \\ \mathbf{\mathcal{L}}[f(at)] = \frac{1}{a}F(\frac{s}{a}) \\ \mathbf{\mathcal{L}}\left[\frac{df}{dt}\right] = sF(s) - f(0-) \\ \mathbf{\mathcal{L}}\left[\int_{0-}^{t} f(\tau)d\tau\right] = \frac{F(s)}{s} \\ \mathbf{\mathcal{L}}\left[\int_{0-}^{t} f(\infty)d\tau\right] \\ \mathbf{\mathcal{L}}\left[\int_{0-}^{t$$

Frequency Shift Time Shift Scaling Differentiation

Integration Theorem

Final Value Theorem

Initial Value Theorem

Key points

- State space model.
- Given a dynamic model, write down its state space model. Example: Eqs. (3-4).
- Laplace transform.

Homeworks

Prove (2), (7), (8).